



Plateau d'Architecture des Systèmes Orbitaux

### Masts for long focal lengths

DEFI study and MA2C R&T Emmanuel Hinglais Christophe Castéras

#### Introduction

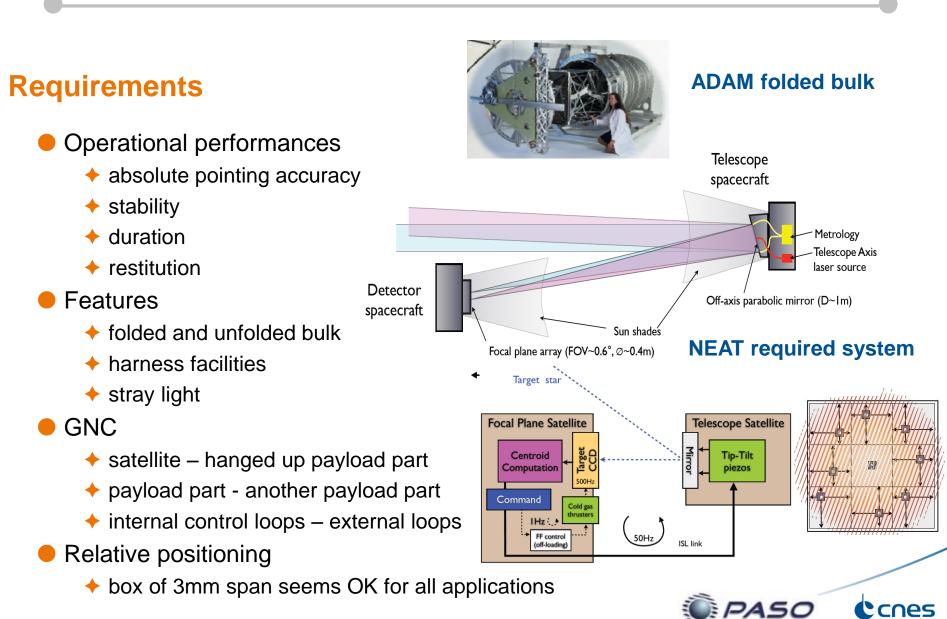
- Space agencies still hesitate to select astrophysics missions using deployable masts or formation flying
- A previous trade off shown that, up to 50m, a deployable mast could be much cheaper to implement than formation flying for such a type of missions
- The US Adam mast has already flown several times with lengths between 10m and 60m for civil applications
  - NuSTAR mast behavior is very convincing
- CNES is currently developing a deployable carbon structural mast that is expected equivalent performances to Adam mast in a passive mode and upper ones in driving mode

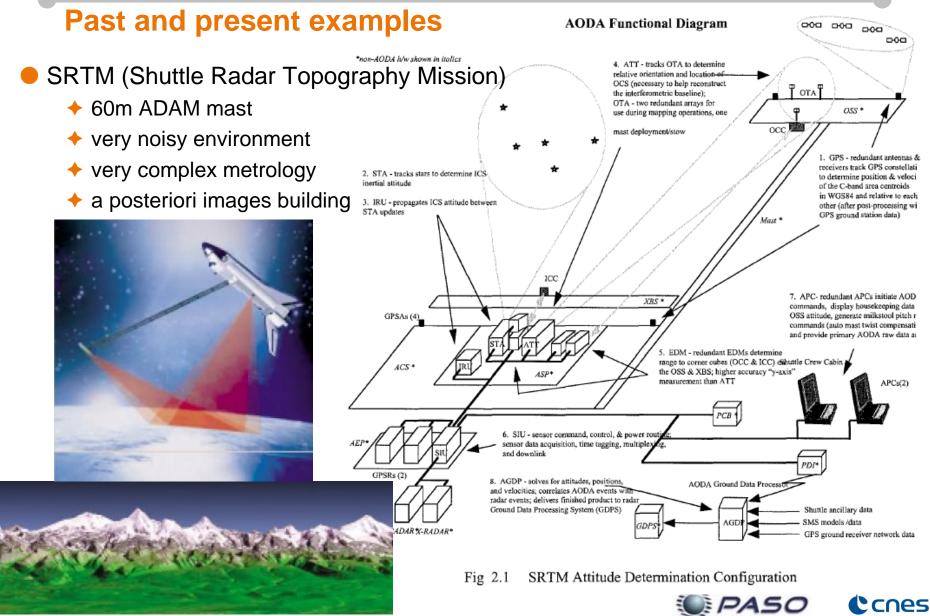




- High energy astrophysics
  - focalizing photons imposes long focal length
    - » Each photon is tagged → image can be re-built a posteriori with the telescope alignment knowledge
  - Allowed misalignment is given by the detection plan oversizing w.r.t the field of view
    - ➔ generally few mm
  - + the frequency bandwidth of troublemakers impacts on front end electronics performances
  - examples: PheniX, COSPIX (ex SIMBOL-X), Dual (γ), ...
- Ultra fine astrometry (NEAT)
  - not foldable focal length avoiding optical defects due to added optical elements
  - observation in visible wavelength with multi short exposure time and averaging
    - ➔ mixt between pointing absolute performance need and averaging using very high performance laser metrology at final stage.
  - ✦ → request additional internal control loops





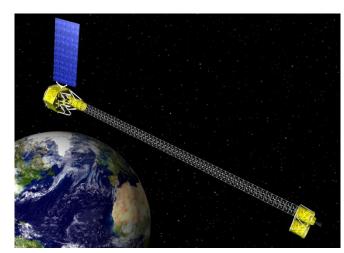


#### Past and present examples

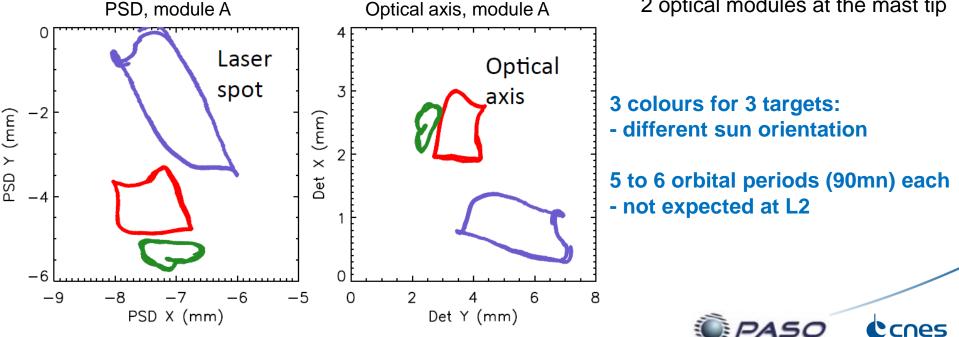
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NuSTAR (Nuclear Spectroscopic Telescope Array)

- Adam mast (ATK): 10m + high inertia on both parts
- + low earth equatorial orbit  $\rightarrow$  several troublemakers
- only thermoelastic effects are measurable



2 optical modules at the mast tip



#### Head mast motion compared to detection plan

## MA2C: Mat Auto-deployable Auto-verrouillable en Carbone (self spread out and self locked carbon mast)

#### Concept (CNES property)

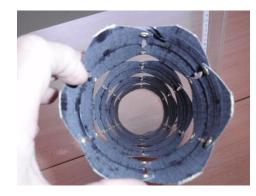
 ultra light cylindrical mast made of independent sections lined up and end to end assembled

section

- » all identical
- » made of a set of plates

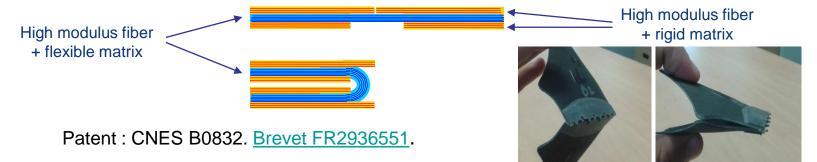


- thin carbon fibre reinforced polymers (CFRP)
- ➢ pivot links and flexible hinges locked once spread out



MA2C: Mat Auto-deployable Auto-verrouillable en Carbone (self spread out and self locked carbon mast)

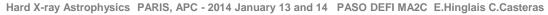
- Technology (CNES property)
  - flexible hinge, locked once spread out
    - » thin carbon fibre reinforced polymers (CFRP)



design features

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- Iow mass and high rigidity
- compact when folded
- intrinsic high mechanical and thermal stability
- wild range of operating and non operating temperatures
- potential high lengths (the section grows with the length)
- high mechanical and thermal stability
- potential static and dynamic behaviour control capability



# MA2C: Mat Auto-deployable Auto-verrouillable en Carbone (self spread out and self locked carbon mast)

CNES R&T with SMEs in Toulouse (Latécoère services, CLIX and CRITT)
 Accommodation (CNES property)

 internal harness
 customizable for specific missions
 folded and spread out dimensions
 static and dynamic behaviour

50 mm

Hard X-ray Astrophysics PARIS, APC - 2014 January 13 and 14 PASO DEFI MA2C E.Hinglais C.Casteras

#### MA2C: CNES R&T present development

- 2m long mast manufacturing then 4m long one end 2014
  - behaviour characterization
    - » deployment test in a large range of temperature [-40°C, + 80°C] in vacuum chamber
    - » Og deployment during parabola flight
    - » static and dynamic behaviour measurement
    - » deployment monitoring system development
    - » end position deployment damping system development
    - » local behaviour understanding
  - linear and non linear finites elements models updating
    - » NASTRAN, FEMAP and SAMCEF
  - same process for 2m mast this year and 4m mast next year
- Objectives
  - having credible mathematical models for project development up to PDR
    - » scaling capacities up to 40m
    - » precise static and dynamic behaviour under different space environments for instrumental and GNC sizing